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Syllabus | Aircraft Stability and Control | Aeronautics ...

6 CHAPTER 1. INTRODUCTION TO FLIGHT DYNAMICS while the aerodynamic moment coefficients are defined as $C_l = L / Q S b$ $C_m = M / Q S c$ $C_n = N / Q S b$ (1.4) Note that the wing span is used as the reference moment arm for the rolling and yawing moments, while the mean aerodynamic chord is used for the pitching moment.

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